

MECH 7710 Homework Assignment #4

(Kalman Filtering)

Note: All problems are to be sampled at 10 Hz ($T_s=0.1$) except for problem #3.

1. Kalman Filter at its best – simulation (actually the Kalman filter is also quite reliable when we have an excellent model and low noise sensors). Suppose we have a 2nd order system that we are regulating about zero (position and velocity) by wrapping an “optimal” control loop around the system. The new dynamics of the continuous time system are given by the closed-loop A matrix:

$$A_{CL} = \begin{bmatrix} 0 & 1 \\ -1 & -1.4 \end{bmatrix}$$

Suppose our measurement is simply position ($C=[1 \ 0]$). There is a white noise process disturbance (force, $B_w=[0 \ 1]^T$) acting on the controlled system.

- a) Simulate the controlled system with the disturbance force ($1\sigma=2$) and a sampled sensor noise ($1\sigma=1$) for 100 seconds at a 10 Hz sample rate.
- b) What is Q , Q_d and R_d ?
- c) Calculate the steady state Kalman gain for the system. This can be done in one of many ways: iterate the kalman filter until it converges, dlqe.m, dare.m, kalman.m, dlqr.m (+ predictor to current estimator trick), etc. What is the steady state covariance of the estimates after the time update, $P^{(-)}$, as well as after the measurement update, $P^{(+)}$. Where are the poles of the estimator?
- d) Now use the steady state kalman filter to generate an estimate (\hat{x} and $\dot{\hat{x}}$) of the 2 states over time. Calculate the norm of the standard deviation of the errors for each state.

$$N = \sqrt{(std(\dot{x} - \dot{\hat{x}}))^2 + (std(x - \hat{x}))^2}$$

- e) Change the ratio of the Q and R_d weights in the Kalman filter design (and repeat part d with the new Kalman gain but DO NOT regenerate a new x and \dot{x}) and determine the effect on the estimation errors. For what ratio of Q to R_d are the errors minimized? Note: Often in practice we do not know the actual Q and R_d so these tend to be “tuning” parameters we can use to tune our filter. However, according to Kalman the estimation errors are only minimized if we use the Q and R_d of the physical system.
2. Download the data **hw4_2** from the website. The data is in the form $[t \ y]$. Suppose we want to design an estimator to estimate the bias in the measurement y . We believe that the bias (x) is constant so we use the model given by:

$$\begin{aligned}\dot{x} &= 0 \\ y_k &= x_k + v_k \\ v_k &\sim N(0,1)\end{aligned}$$

- a) Run the Kalman filter estimator with $Q_d=0$. What happens at $t>100$ seconds? Why? Calculate the steady state Kalman gain L_{ss} . Plot $L(k)$. This is known as the filter “going to sleep” (becomes a least squares estimator).
- b) To offset this problem we will “tune” Q_d to track the bias. What is the effect of changing Q_d on the ability to track the step change in the bias? (try values of Q_d from 0.0001 to 0.01 and plot $L(k)$ as well as the estimate of the bias (\hat{x}). What is the tradeoff?
- c) Now filter the measurement using the first order low-pass filter

$$H(z) = \frac{\sqrt{Q_d}}{z - (1 - \sqrt{Q_d})}$$

- use the command: `yf=filter(numd,dend,y,y0)`
- d) How does this compare to the Kalman filter solution. Why are these two filters the same for this problem?

3. Design a “Navigation” type Kalman filter to estimate the states [East, North, Radar_Bias, Psi, Gyro_Bias]. (Note: this is a non-linear problem that requires an Extended Kalman Filter (EKF) to do correctly. However we can solve the problem in one of two ways: (i) linearize the equations about the nominal operating point and produce a constant A matrix for that operating point, or (ii) simply update the A matrix at every time step with our measurements or estimates. Download the data **hw4_3** from the website and run the filter sampled at 5 Hz.

- a) How did you choose the covariance values for Q_d (especially for the radar and gyro biases).
- b) How does the bias estimation compare to a Least Squares Solution. How does the bias estimate compare to the Recursive Least Squares solution if you make the covariance (Q_d) of the bias estimates equal to zero.
- c) Integrate the last 40 seconds of data to see how well you have estimated the biases. This can simply be done by “turning off” the measurements in the observation matrix! Why do the bias estimates remain constant during the 40 second “outage?”

4. Estimator for Vehicle Dynamics. The yaw dynamics of a car (for a stability control system) can be described by the following model (at 25 m/s):

$$\dot{X} = \begin{bmatrix} -2.62 & 12 \\ -0.96 & -2 \end{bmatrix} \begin{bmatrix} \dot{\psi} \\ \beta \end{bmatrix} + \begin{bmatrix} 14 \\ 1 \end{bmatrix} \delta$$

$$y_k = [1 \quad 0] X_k + v_k$$

where:

$$\begin{aligned}\dot{\psi} &= \text{Vehicle Yaw Rate} \\ \beta &= \text{Vehicle SideSlip Angle} \\ \delta &= \text{Steer Angle} \\ v_k &= \text{Sample Sensor Noise}\end{aligned}$$

- a) Assuming we can only measure the yaw rate ($v_k \sim N[0, (0.1)^2]$), design a Kalman filter to do full state estimation (select a reasonable Q_d). Provide a unit step steer input and estimate both states. On one page plot the actual states and estimated states (use subplot(2,2,n) for each of the two states). Where are the steady state poles of the estimator?

Now, somebody has loaded the trunk of the vehicle with bricks, changing the CG of the vehicle so now the actual model (at 25 m/s) is:

$$\dot{X} = \begin{bmatrix} -2.42 & 4 \\ -0.99 & -2 \end{bmatrix} \begin{bmatrix} \dot{\psi} \\ \beta \end{bmatrix} + \begin{bmatrix} 18 \\ 1 \end{bmatrix} \delta$$

$$y_k = \begin{bmatrix} 1 & 0 \end{bmatrix} X_k + v_k$$

NOTE: We do not know that somebody has loaded the trunk and that the C.G. has changed, therefore we must use the model for part a in our Kalman Filter.

- b) Redo part a.. Can you estimate the slip angle correctly? Try various Q_d .
 c) Now lets say we have a noisy measurement of the slip angle ($\eta_k \sim N[0, (0.5)^2]$):

$$y_k = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} X_k + \begin{bmatrix} v_k \\ \eta_k \end{bmatrix}$$

Assuming the sensor noises are uncorrelated, what is R?

- d) Redo part a. What is the effect of changing the element of Q_d associated with the slip angle estimate. What must Q_d equal to ensure an unbiased estimate of the states (How much filtering does that provide)?

5. **(OPTIONAL)** *Estimator of a non-minimum phase system.* Consider the system dynamics:

$$\begin{aligned} \dot{x}_1 &= w \\ \dot{x}_2 &= -x_1 - 2x_2 + w \end{aligned}$$

with continuous measurements $y(t) = x_2(t) + v(t)$. The measurements and process noise are both independent white noise processes (spectral densities R_v and R_w respectively).

- a) Find the transfer function $G_{yw}(s)$ from $w(s)$ to $y(s)$ and identify the poles and zeros. Recall that right-half place zeros are called non-minimum phase.
 b) Use this transfer function to sketch the symmetric root locus for the time-invariant estimator poles vs. R_w/R_v . Clearly demonstrate that one of the estimator poles tends towards $s = -1$ as $R_v/R_w \rightarrow 0$, even though there is no zero there (only a reflected one). Confirm that this implies that part of the estimation error will never attenuate faster than e^{-t} even with noise-free measurements.
 c) Use the solution of the steady-state Riccati equation to show that as $R_v/R_w \rightarrow 0$, the

$$P \rightarrow \begin{bmatrix} 2R_w & -\sqrt{R_w R_v} \\ -\sqrt{R_w R_v} & \sqrt{R_w R_v} \end{bmatrix}$$

- d) Use this result to confirm that x_1 can not be estimated perfectly, even with $R_v \rightarrow 0$.
 e) Comment on the impact of the non-minimum phase zero on our ability to estimate this system's state.