

1. Mode-Matching Effects

def: In MEMS gyros, mode-matching refers matching the drive and sense resonant frequencies:

$$\omega_d = \omega_s$$

Let's consider the effects

a. Drive Side

$$m_d \ddot{x} + c \dot{x} + kx = A_x \sin(\omega_d t)$$

solving for $x(t)$ { ELEC 5760/6760 }

$$x(t) = -\frac{A_x}{c \omega_d} \cos(\omega_d t)$$

$= -X_0 \cos(\omega_d t)$ → assumes drive side control system
(i.e. "resonator") keeps X_0 constant

$$\therefore \dot{x}(t) = X_0 \omega_d \sin(\omega_d t)$$

b. Sense Side

$$\text{Coriolis Force} = F_c = 2m_d \Omega_L \dot{x} = 2m_d X_0 \omega_d \Omega_L \sin(\omega_d t)$$

$$\therefore m_s \ddot{y} + c \dot{y} + k_y + F_c = 0$$

→ This assumes $m_s = m_d$ or $m_s \neq m_d$

$$\therefore m_s \ddot{y} + c \dot{y} + k_y = -2m_d X_0 \omega_d \Omega_L \sin(\omega_d t)$$

The solution is: $y(t) = Y_0 \cos(\omega_d t + \phi_s)$

$$\text{where: } Y_0 = \Omega_L \frac{m_d \omega_d}{m_s \omega_s^2} \left(\frac{2X_0}{\sqrt{\left[1 - \left(\frac{\omega_d}{\omega_s}\right)^2\right]^2 + \left[\frac{1}{Q_s} \left(\frac{\omega_d}{\omega_s}\right)\right]^2}} \right)$$

$$\phi_s = -\tan^{-1} \left(\frac{\frac{1}{Q_s} \left(\frac{\omega_d}{\omega_s}\right)}{1 - \left(\frac{\omega_d}{\omega_s}\right)^2} \right)$$

$$\text{where: } \omega_s = \sqrt{\frac{k}{m_s}} \text{ and } Q_s = \frac{m_s \omega_s}{c}$$

① if $\omega_s = \omega_d$, i.e. mode-matched,

$$\text{then } y_0 = \Omega \frac{2Q_s X_0 M_d}{m_s \omega_s}$$

and if $m_s = m_d = m$,

$$\text{then } y_0 = \frac{\Omega 2 X_0}{c} = \frac{2 A_x m}{c^2 \omega_d}$$

and $\phi_s = -90^\circ$

$$\therefore y(t) = \frac{2 A_x m \Omega}{c^2 \omega_d} \cos(\omega_d t - 90^\circ)$$

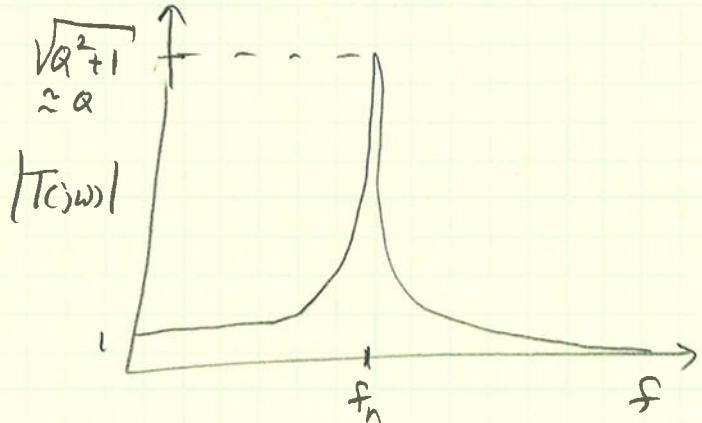
$$= \frac{2 A_x m \Omega}{c^2 \omega_d} \sin(\omega_d t) \rightarrow \text{same as in ELEC5760/6760}$$

② what if $\omega_s \neq \omega_d$, maybe due to fabrication errors

Consider a typical

MEMS resonator:

high Q



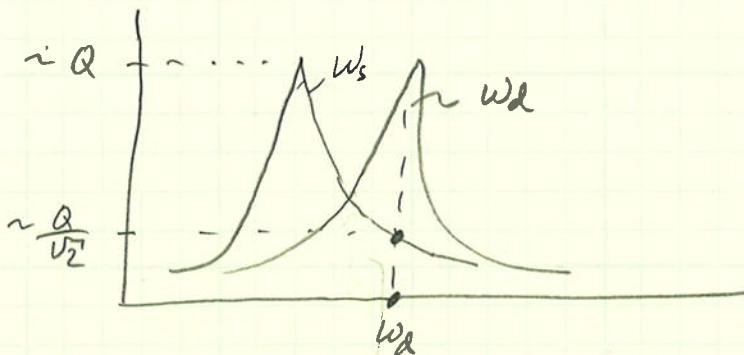
$$3\text{dB BW} \approx \frac{f_n}{Q}$$

Assume $Q = 10,000$ and $f_n = 10\text{kHz}$

$$\therefore \text{BW} = 1\text{Hz}$$

so being off just $1/2\text{Hz}$, and the gain drops by $\frac{1}{2}$
or 29.3%

So if ω_s and ω_d are off by $\gamma_2 \text{BW}$ (0.5Hz here)



y_0 becomes: $y_0 \approx \mathcal{L} \frac{2 Q_s X_0 M_d}{\sqrt{\sum M_s} \omega_d} \rightarrow$ results in gain error
and $\phi_s \neq -90^\circ \rightarrow$ problem for synchronous
demodulation circuit
using $U_x \sin(\omega_d t)$

③ Assume perfect mode matching: $\omega_d = \omega_s$

Let \mathcal{L} not be constant

$$\mathcal{L}(t) = \mathcal{L}_0 \sin(\omega_r t) \text{ where } \omega_r \ll \omega_d$$

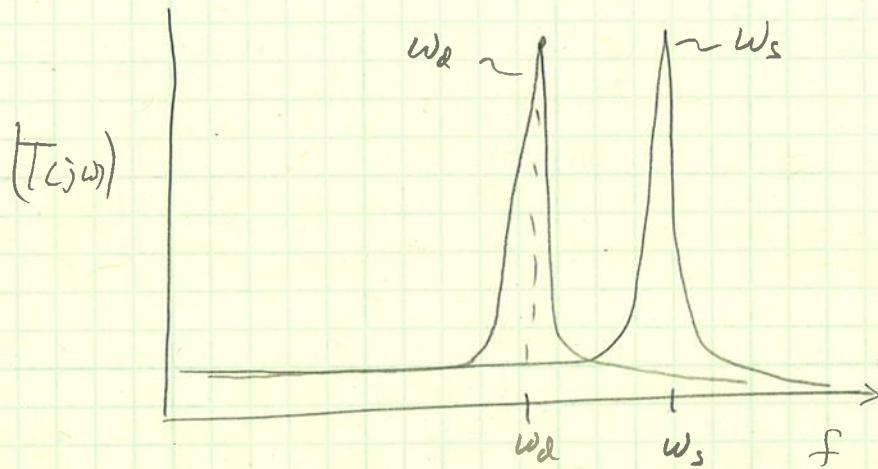
$$\begin{aligned} \text{i. Coriolis force: } F_c &= 2 m_d \mathcal{L}(t) \dot{x} \\ &= 2 m_d \mathcal{L}_0 \sin(\omega_r t) X_0 \omega_d \sin(\omega_d t) \\ &= \mathcal{L}_0 m_d X_0 \omega_d (\cos(\omega_d - \omega_r) - \cos(\omega_d + \omega_r)) \end{aligned}$$

Frequency of $F_c \neq \omega_d$

\rightarrow same effect as if $\omega_s \neq \omega_d$

2. A Solution

choose $\omega_s > \omega_d$ for high Q resonators



Now, the gain is lower (much lower) on the sense side, but the change in gain for Δf about f_d is small.

→ can better tolerate fabrication tolerances

→ can handle time varying rotations

gyro BW typically defined as $f_s - f_d$

Note, still want to closely control Q through C

→ want to maintain constant, known pressure around device in a hermetic package

→ and measure temperature to calibrate for temperature induced C (Q) changes